

OPUS: Orbital Debris Propagators Unified with Economic Systems

An Integrated Assessment Model of the Orbital
Environment

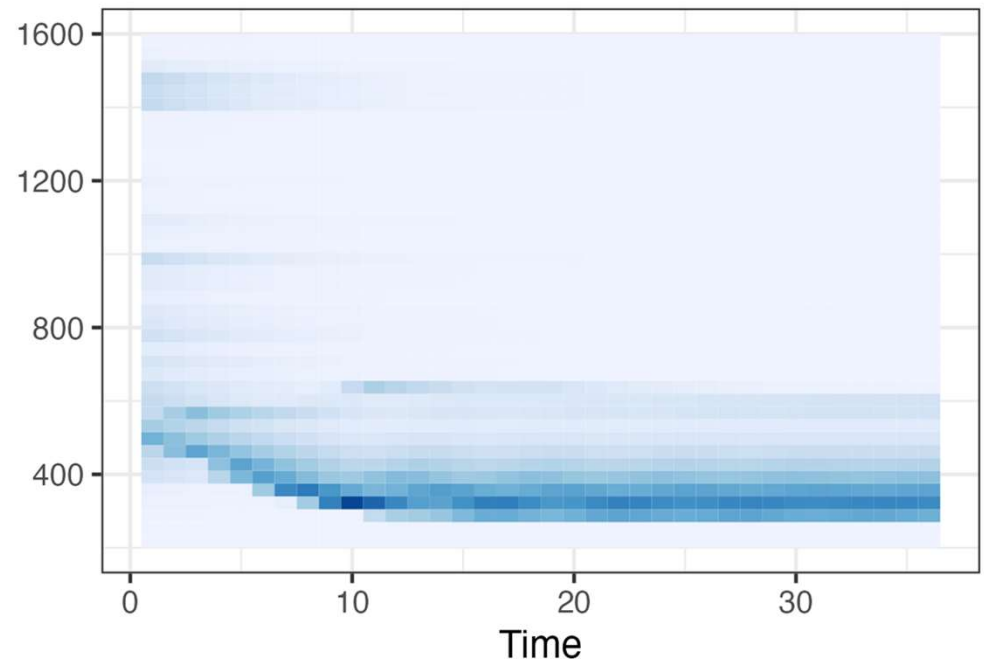
Akhil Rao, Mark Moretto, Marcus Holzinger
Daniel Kaffine, Brian Weeden

Background

- Radical shift underway in how we use space, primarily through the launching of large constellations and many more satellites with shorter lifespans
- Lots of policy proposals being offered to try and solve this issue
 - Technical (reduction in 25-yr rule, Pc vs Time/Area Product)
 - Regulatory (no propulsion above 400 km, mandatory propulsion)
 - Economic (space debris bonds, orbit use fees)
- But lack of tools for policymakers to evaluate the efficacy of these proposals, and in particular how they could impact behavior
 - Many other fields have seen counterproductive outcomes as a result of unexpected changes in behavior to policy initiatives

OPUS: Orbital Debris Propagators Unified with Economic Systems

- Model architecture
 - I/O
 - Debris environment models
 - Economic behavior models
 - Revenues and costs
- Model validation
 - Emergent behaviors
 - Metrics
 - GMPHD
- Policy exercises
 - 25- vs 5-year disposal rule
 - 5-year disposal vs 25-year disposal with OUF
- Future R&D directions



Model architecture

User Inputs:

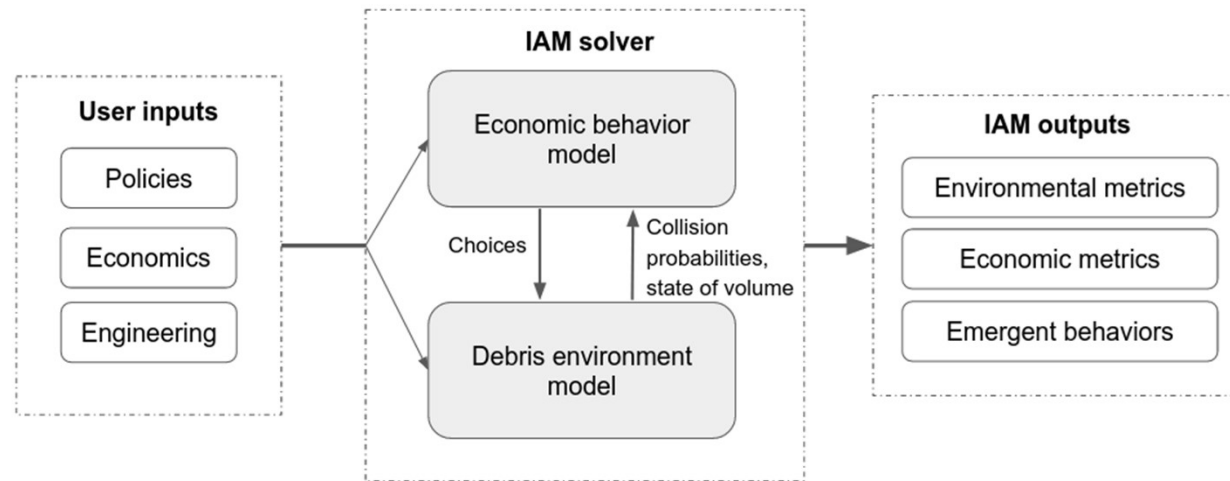
- CSV inputs, bash script

IAM solver:

- Environment model/propagator:
 - MOCAT-4S (main)
 - GMPHD (development)
- Economic model
 - Open-access launching
- MATLAB files: CSV outputs

IAM outputs:

- R files: images, CSV outputs



MOCAT-4S (MIT Orbital Capacity Assessment Tool 4S)

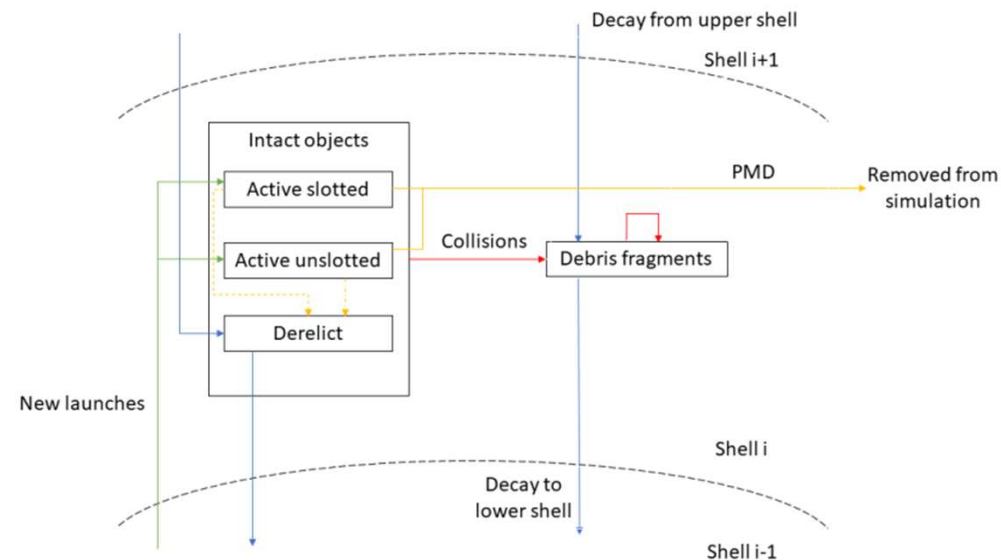
4-species source-sink evolutionary model (SSEM)

- Slotted = Constellation (exogenous)
- Unslotted = Fringe (endogenous)
- Derelicts = big intacts
- Debris = small fragments

Orbital locations parameterized as “shells”

- 40 shells, 200-1600 km altitude
- Each shell 35 km thick
- Annual time steps

Each species has a fixed size/mass



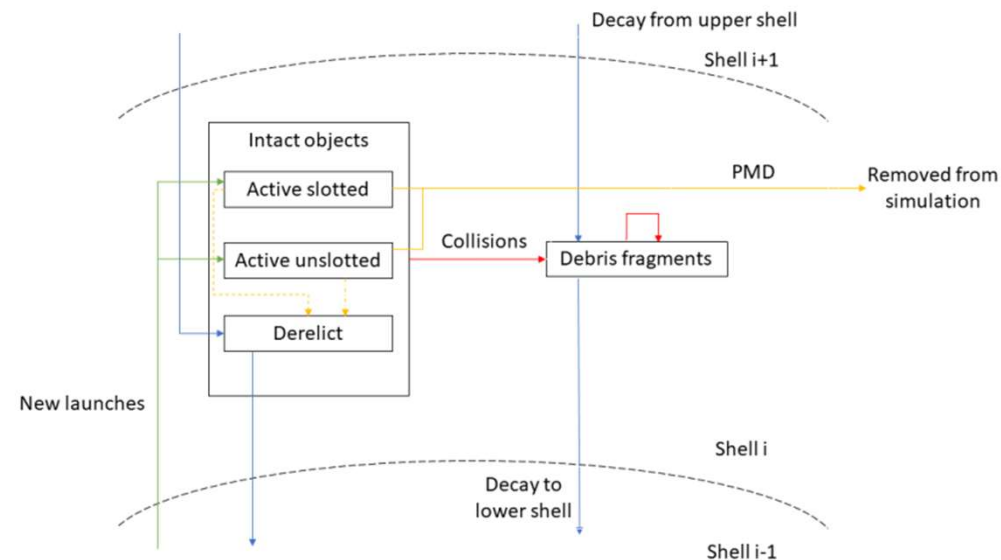
D'Ambrosio, Andrea, Miles Lifson, Daniel Jang, Celina Pasiecznik, and Richard Linares. "Projected Orbital Demand and LEO Environmental Capacity." (2022).

MOCAT-4S (MIT Orbital Capacity Assessment Tool 4S)

MOCAT-4S is the default propagator, integration is most-thoroughly tested

Modifications from baseline state:

- End-of-life disposal produces derelicts at highest naturally-compliant altitude
 - Only for satellites outside naturally-compliant zones
- “Disposal success rate” recast as “disposal non-compliance rate”
 - Default set to “full compliance”



D'Ambrosio, Andrea, Miles Lifson, Daniel Jang, Celina Pasiecznik, and Richard Linares. "Projected Orbital Demand and LEO Environmental Capacity." (2022).

GMPHD (Gaussian Mixture Particle Hypothesis Density)

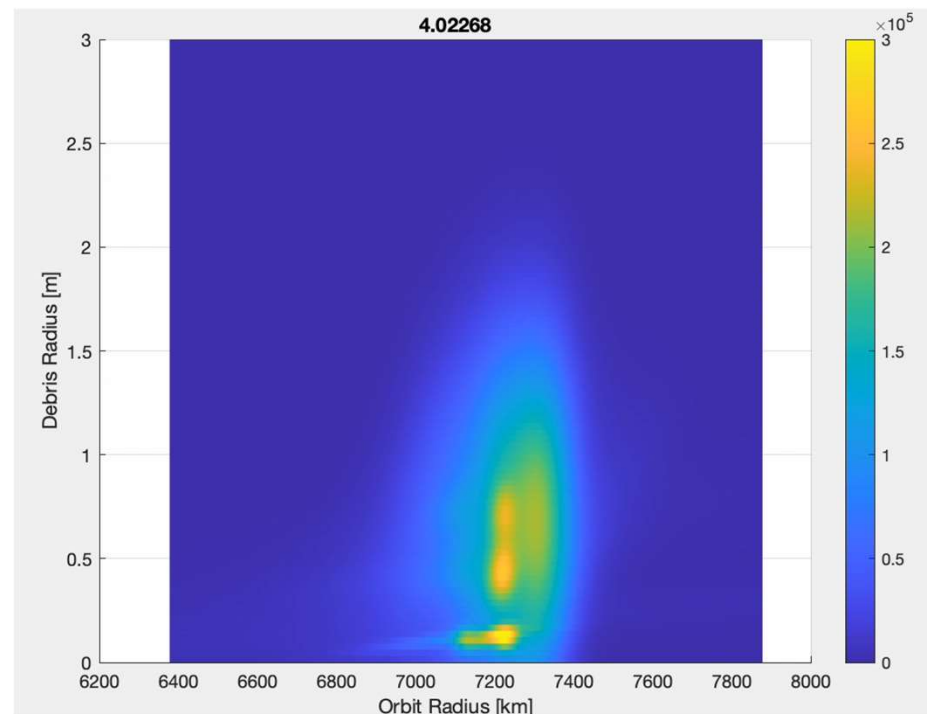
“2+∞”-species SSEM

- Slotted = Constellation (exogenous)
- Unslotted = Fringe (adaptive)
- Debris = all intact+fragments

Continuous locations and debris sizes

- Calculate metrics in discrete “shell-bins”
- Annual time steps
- Can extend to arbitrary continuous indexing (inclination, RAAN, etc)

Debris have evolving distribution of sizes

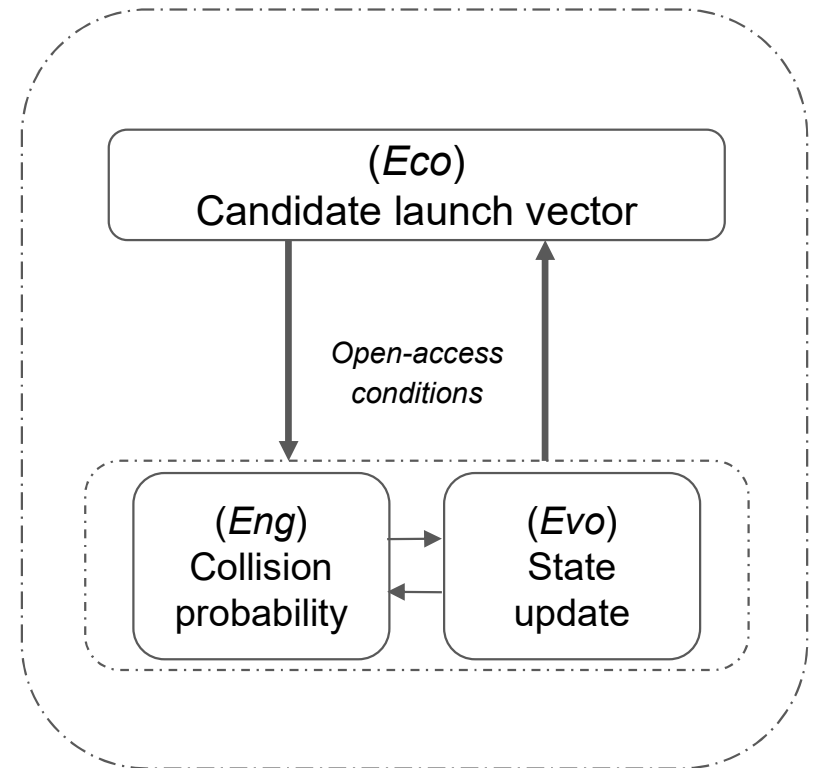


This is a new thing: show our framework is propagator-agnostic + plant seeds for future work + soft pitch on this extension

Economic behavior model

Two types of satellites:

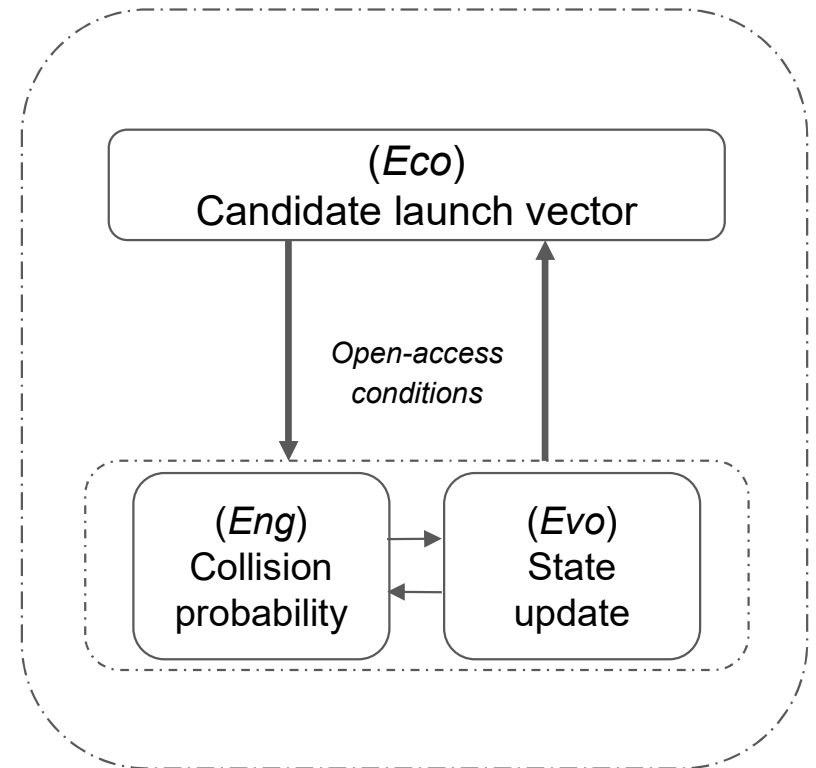
- Constellation
 - Many satellites under one operator
 - One operator per location
 - User-set locations, max launch rates, and target sizes
 - “Slotted”
- Open-access fringe
 - One satellite per operator
 - Many satellites per location
 - Endogenous locations and launch rates
 - “Unslotted”



Economic behavior (Fringe)

Open-access launching:

1. Economic actors make choices
 - a. Launch if positive economic returns anticipated under “true” propagator
 - b. Economic returns:
Rate of return - discount rate - depreciation - collision chance - taxes
 - c. Equilibrium: zero economic profits everywhere
2. Propagate state
3. Repeat



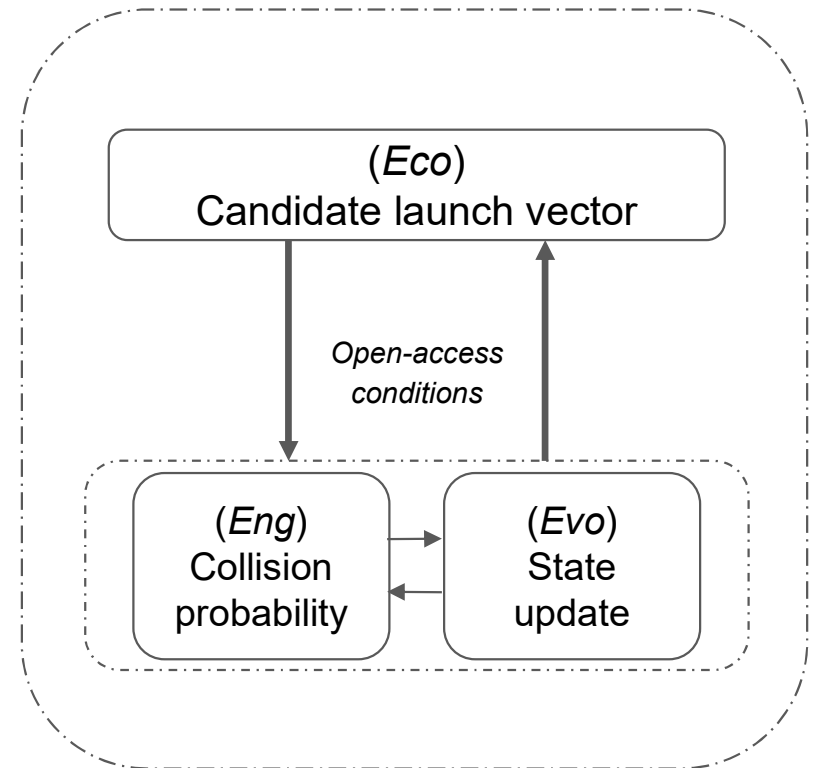
Economic behavior (Fringe)

Open-access launching:

1. Economic actors make choices
2. Propagate state
3. Repeat

Null model: satellite feedback launching:

- Launch rate to location k at time t proportional to satellites in k at time $t-1$
- Proportionality is location-specific: replenish natural decay only



Economic behavior (Constellations)

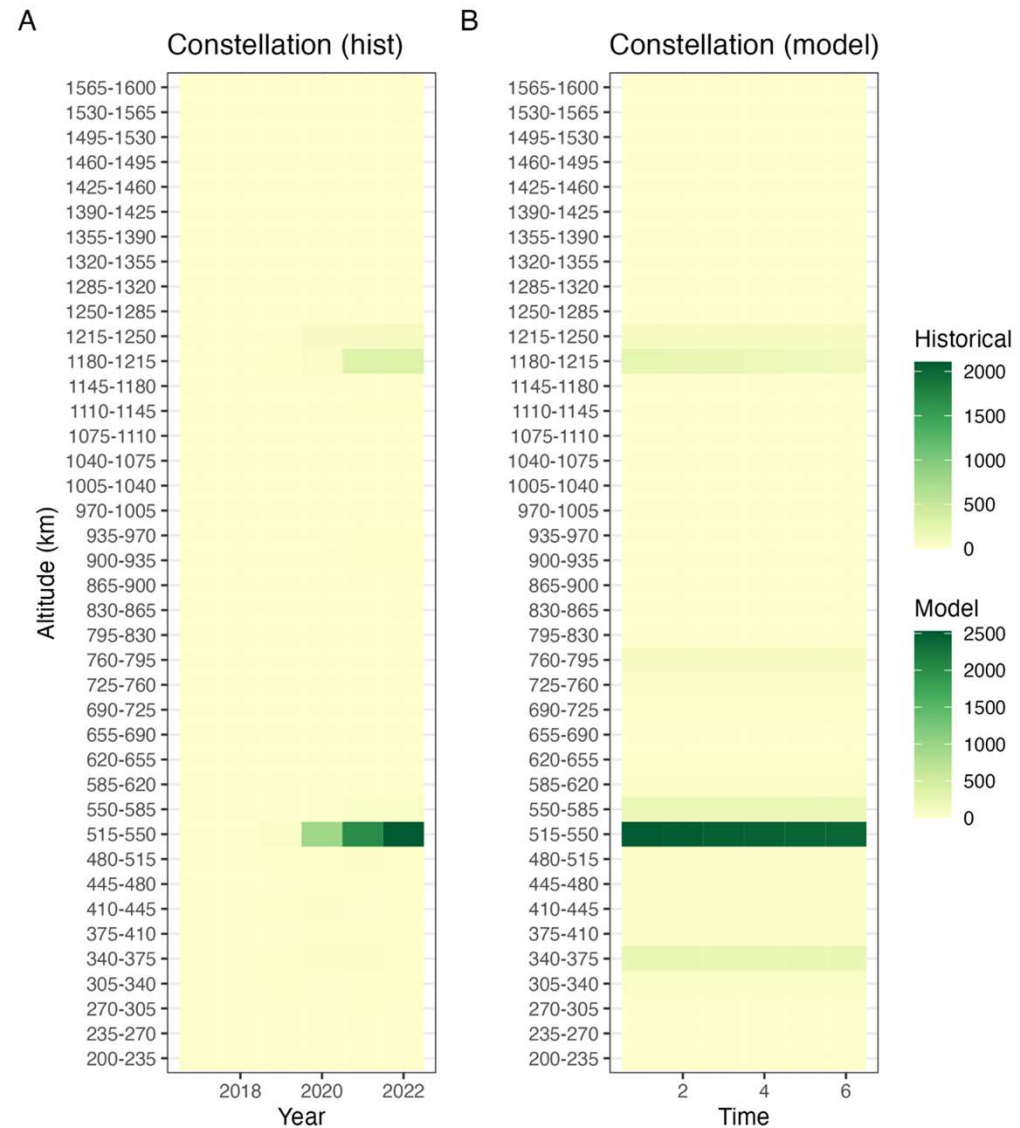
Constellation behaviors are set exogenously

User defines:

- The number of constellations
- Their locations
- Their maximum launch rates
- Their target sizes

Constellations build up to target size as fast as possible and maintain

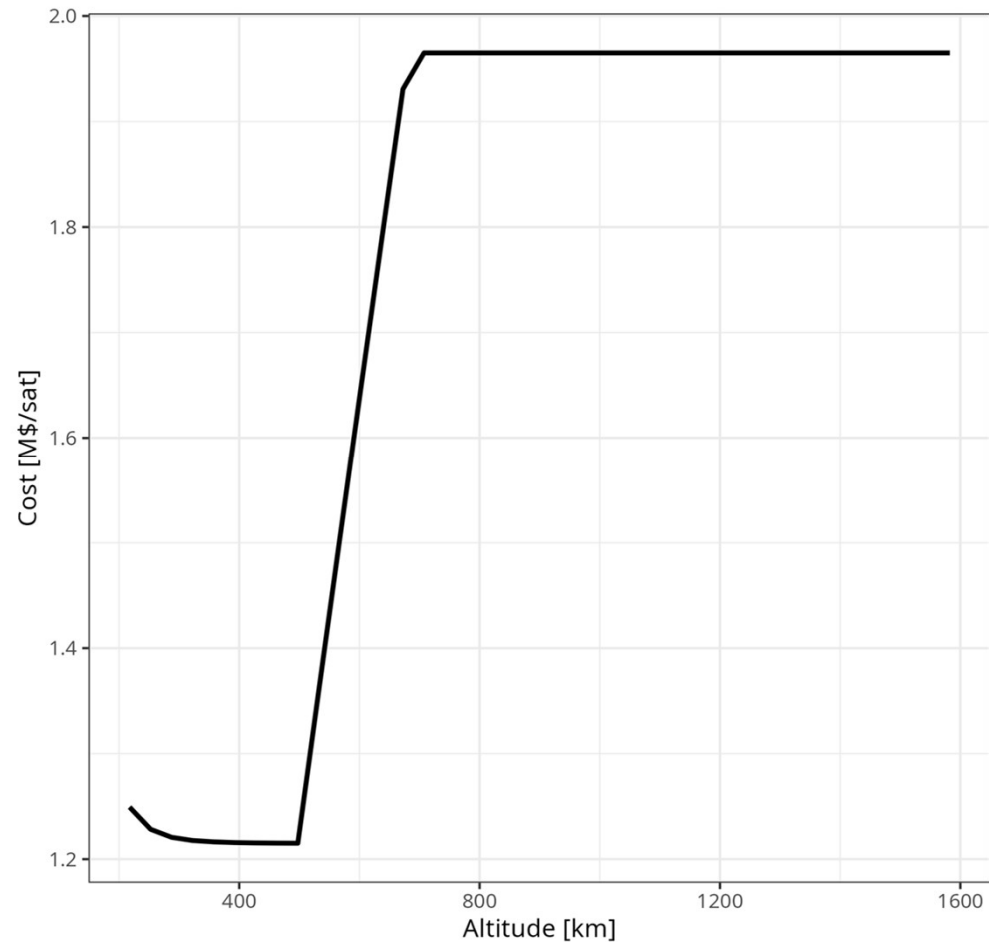
Same average compliance rate as fringe



Fringe costs and revenues

Revenue and costs:

- **Costs:**
 - Lift price: 5000 \$/kg
 - Delta-v cost: 1000 \$/m/s
 - Opportunity cost: forgone revenues from deorbit burns if compliant
- **Revenues:**
 - Linear inverse demand
 - Maximum: 750k \$/year
 - Competition: -100 \$/sat/year
- **Benchmark:**
 - No Pigouvian tax
 - 5-year lifetime (20% depreciation/year)\
 - Discount rate 5%



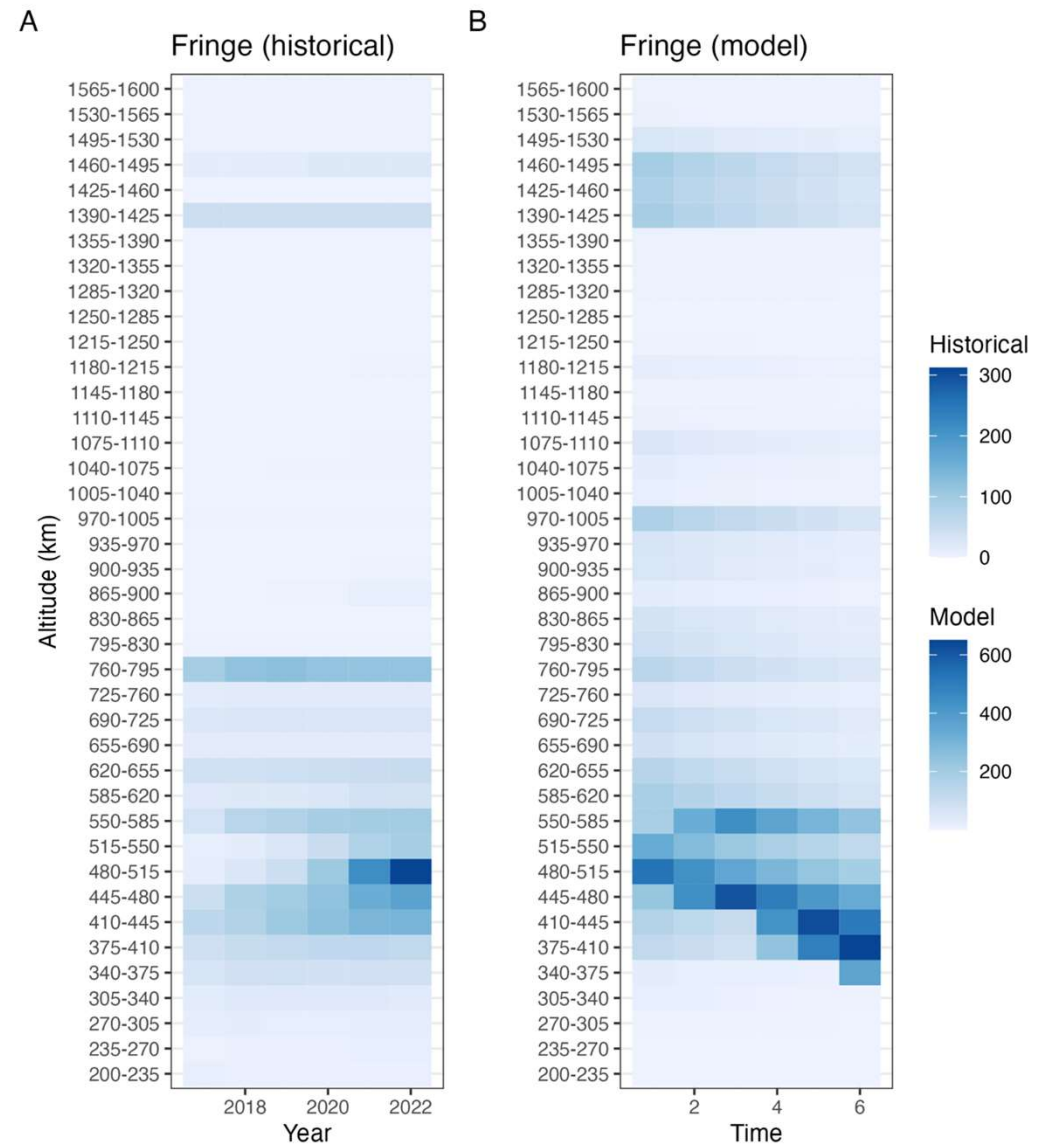
Model validation

Model predictions with 25-year disposal vs historical data 2017-2022

Model captures:

- Prefer naturally-compliant altitudes
- Avoid larger constellation
 - Prefer to be below it
- Avoidance of lowest shells
 - Highest drag

All “emergent” features!

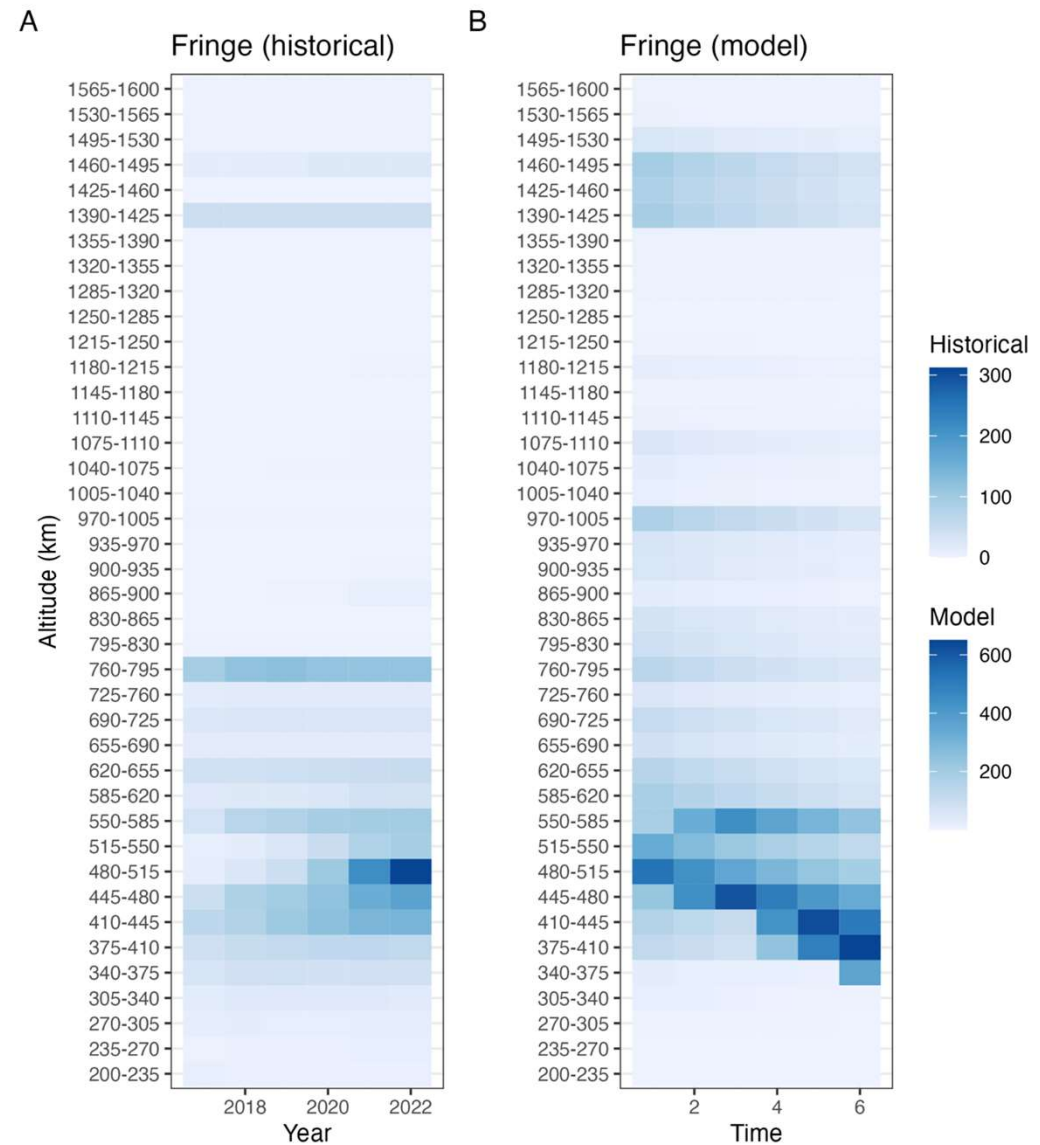


Model validation

Model predictions with 25-year disposal vs historical data 2017-2022

Caveats:

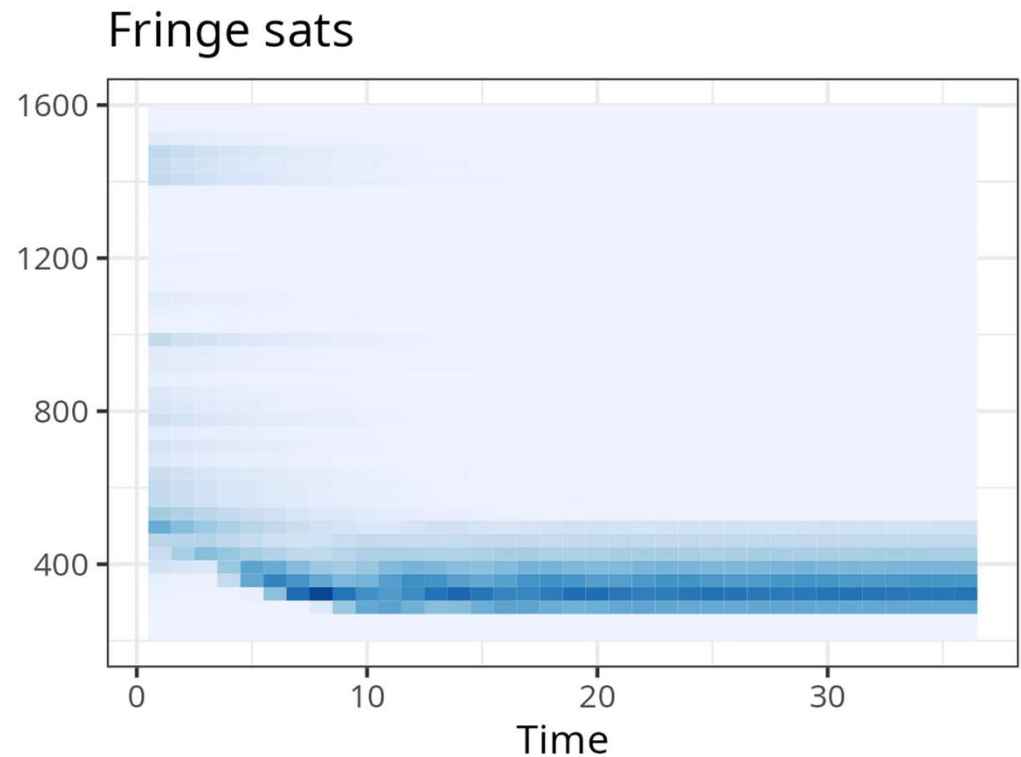
- Declining launch costs (real) vs static launch costs (model)
- Uncalibrated economic parameters
- Model starts from July 2022 population
- Model assumes full compliance
- Constellations start ~2020 (real) vs t=1 (model)



Emergent behaviors (5-year disposal, full compliance)

Some unexpected/plausible behaviors:

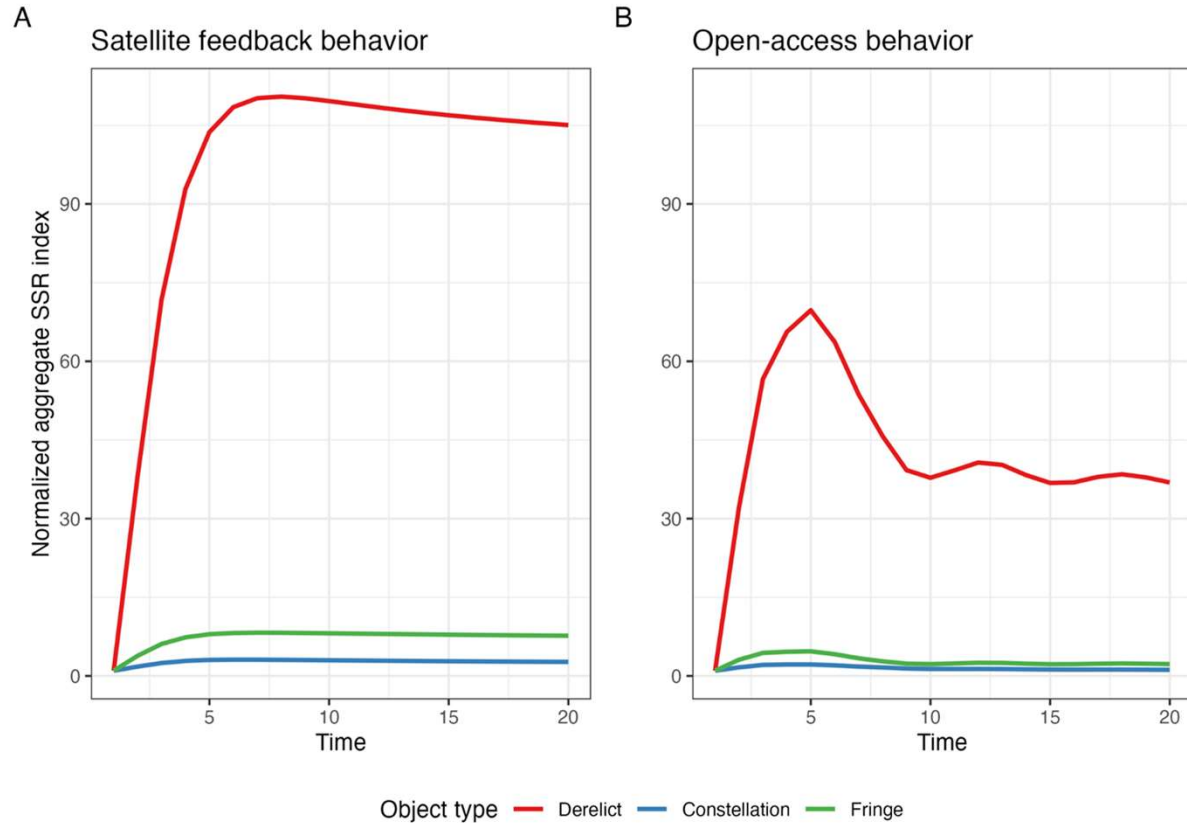
- Periodicity
 - Derelict decay + collective responses => damped oscillations in location values
- Minimum viable altitude
 - Stationkeeping costs => “floor” on how low it’s worth going
- Abandon higher altitudes
 - Too costly to dispose above naturally-compliant region



Metrics: SSR

Can calculate metrics to compare policies / illustrate effects of open-access behavior

- Peak as constellations build up derelicts
- Adaptive responses to collision risk => lower SSR index values (more sustainable)
- Still not maximizing social welfare (unpriced externalities)

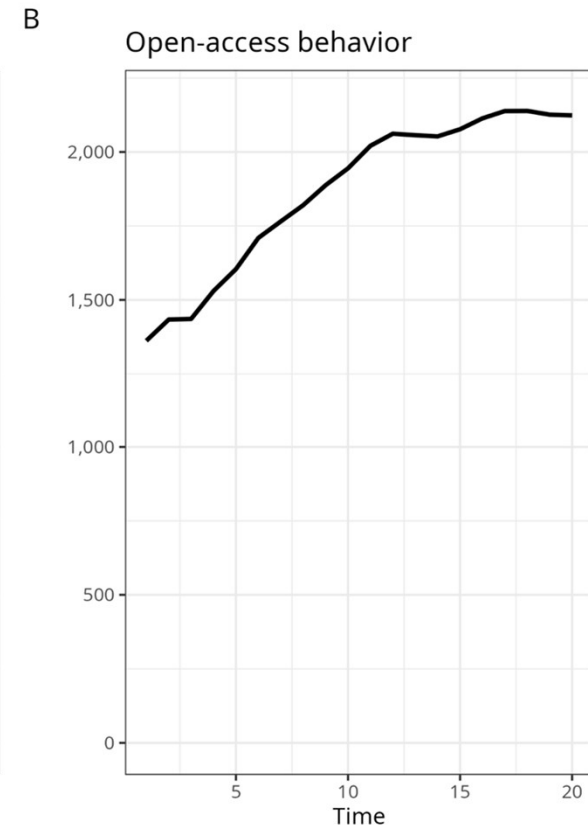
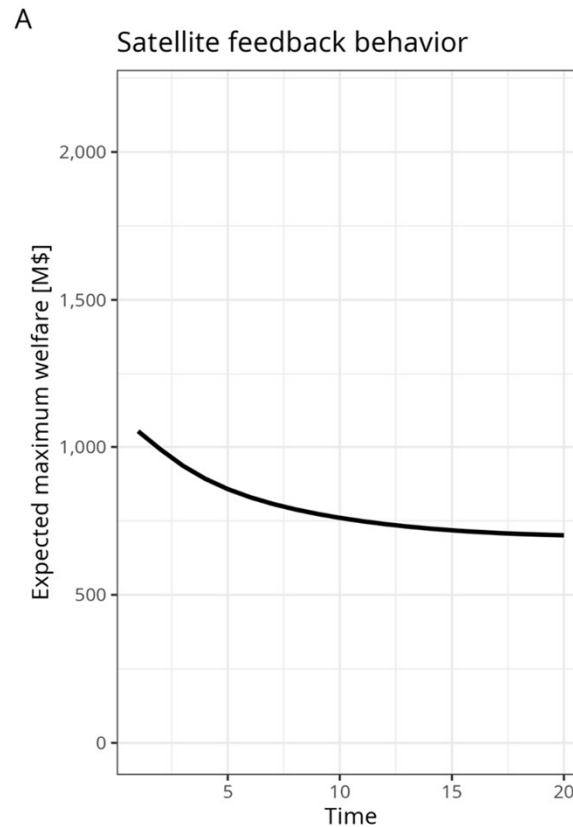


Model is uncalibrated – illustrative only

Metrics: Economic welfare

Can calculate metrics to compare policies / illustrate effects of open-access behavior

- Trough as constellations build up derelicts
- Open-access welfare increasing as firms adapt to environment
- Still not maximizing social welfare (unpriced externalities)



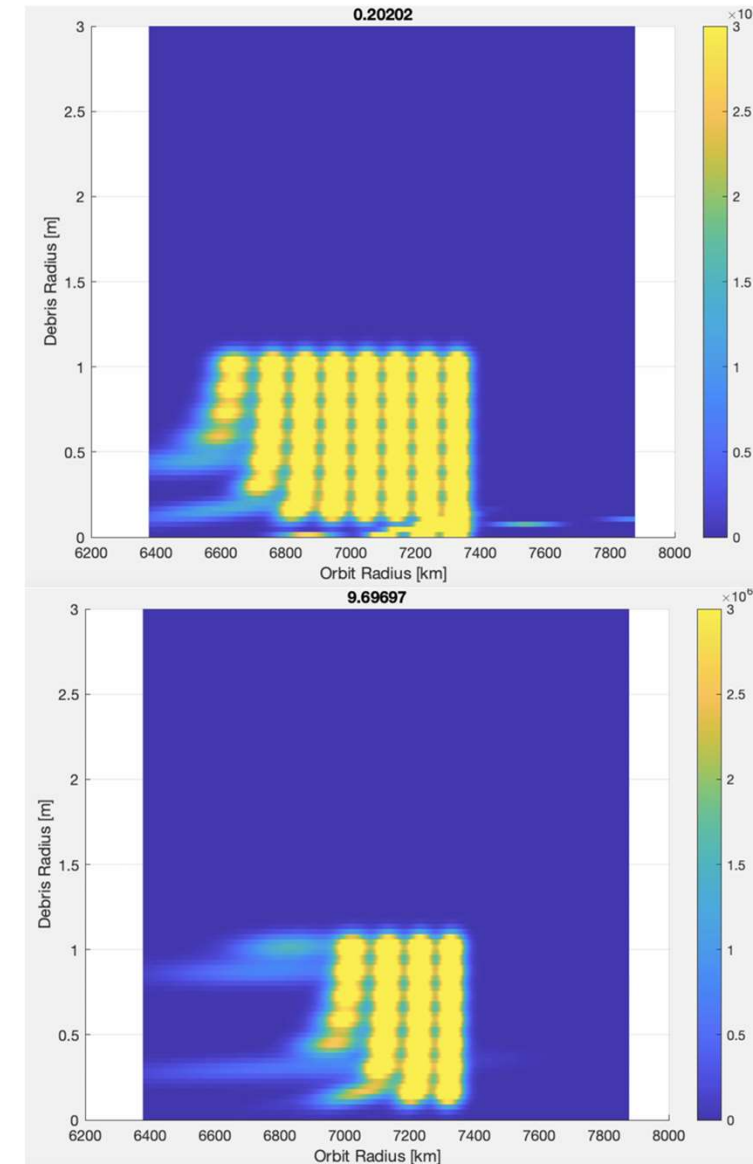
Model is uncalibrated – illustrative only

GMPHD

Currently in proof-of-concept stage

Requires further R&D to develop:

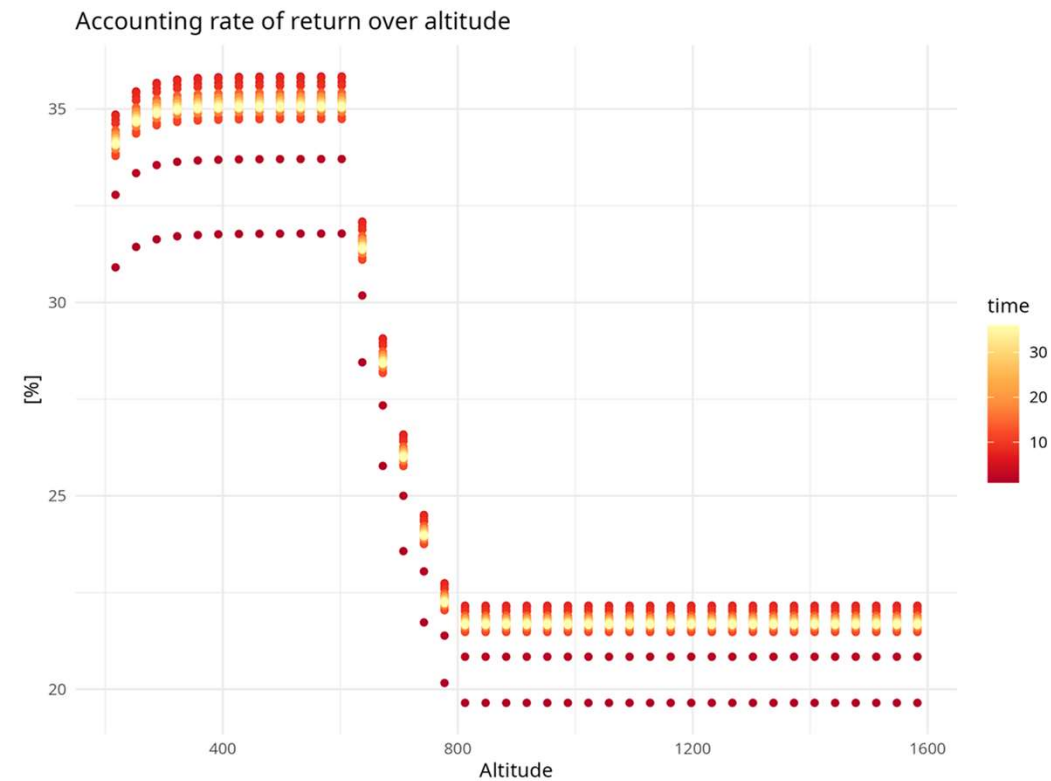
- Model calibration/data constraints
 - Size/altitude distribution of fragments per collision/fragmentation event
 - Minimum lethal sizes across altitudes/velocities
 - Effects of repeated non-lethal impacts across size distribution
- Useful hyperparameter settings
 - Number of Gaussians in mixture
 - Gaussian splitting/merging rates



Policy exercise: 25- vs 5-year PMD

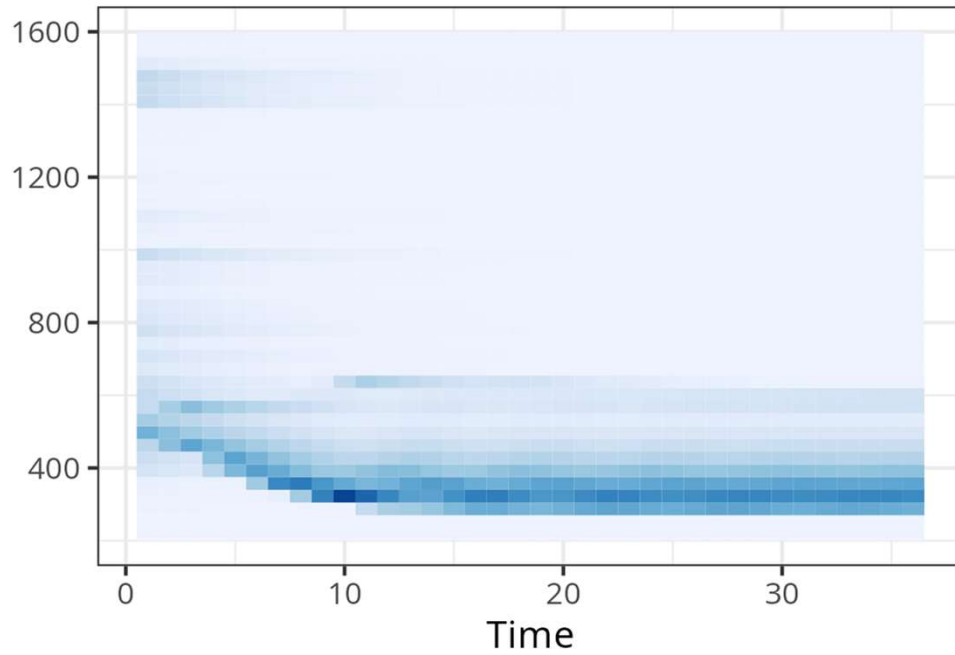
How much better is 25 vs 5 years PMD, accounting for behavior? Can simulate in OPUS

- Two separate scenarios
- Start from July 2022 initial population
- Assume full compliance

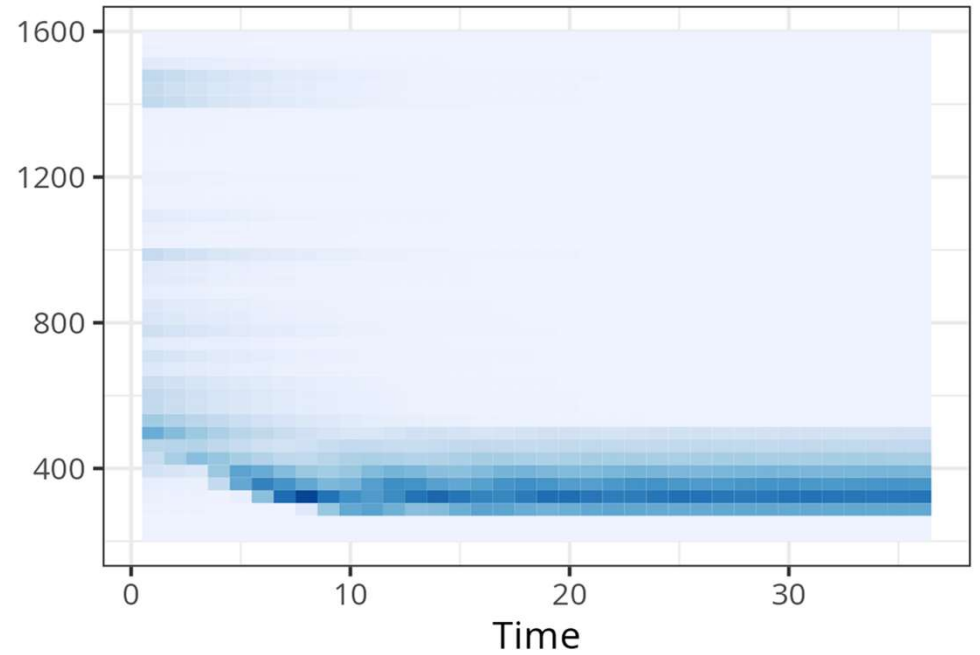


25- vs 5-year PMD (Fringe sats)

25 year PMD



5 year PMD

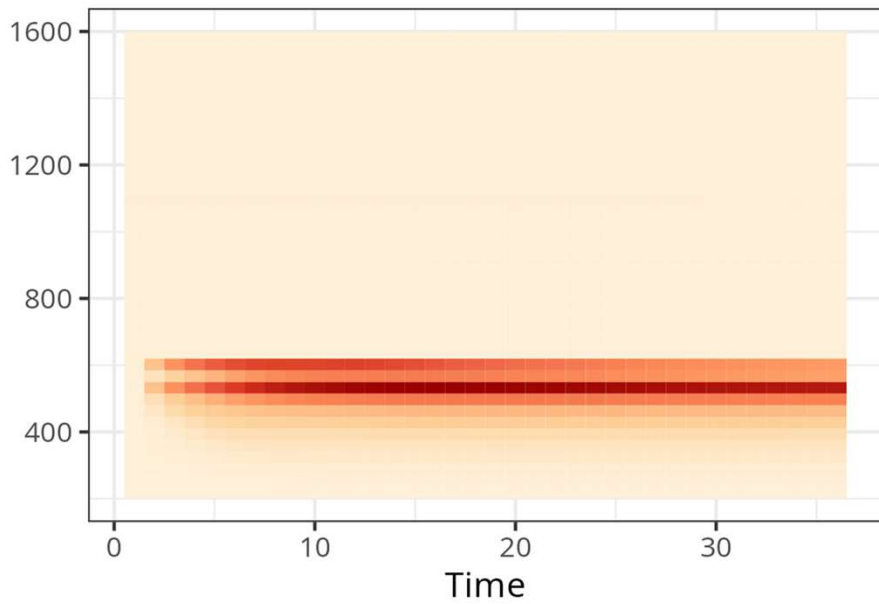


25- vs 5-year PMD (Derelict sats)

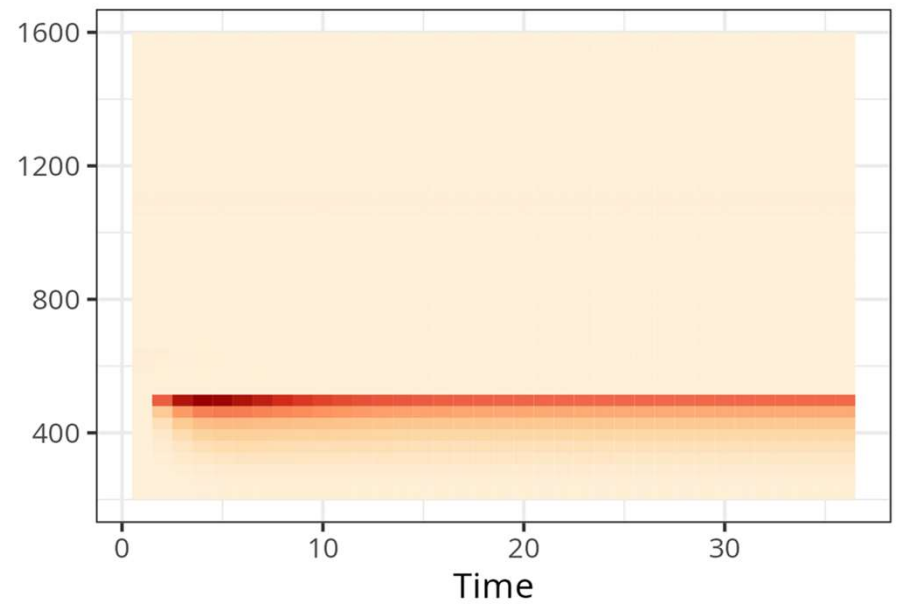
25 year PMD

5 year PMD

Derelict sats



Derelict sats

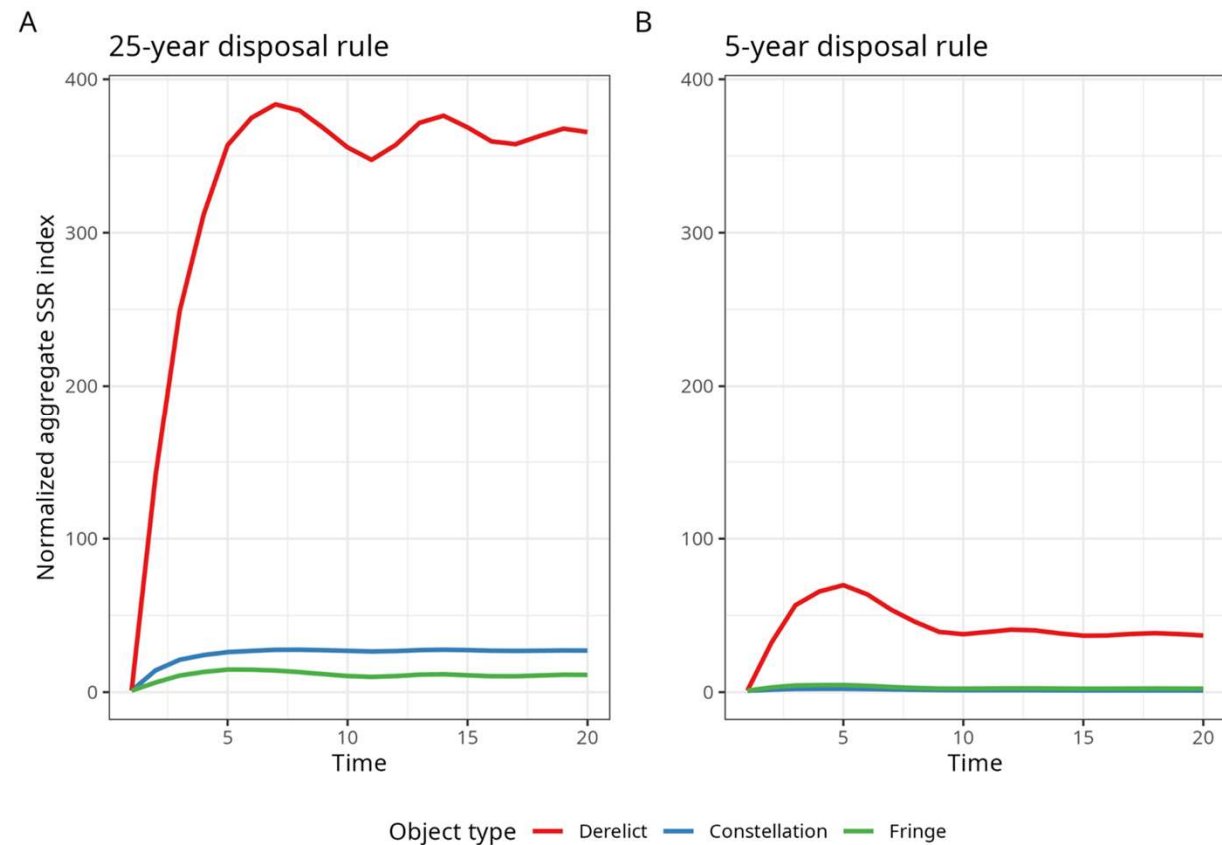


25- vs 5-year PMD (SSR)

Model is uncalibrated – illustrative only

“Aggregate” Space Sustainability Rating (SSR) to assess overall environment state

- Compute SSR across all objects of each type
- Normalize to 1 in initial state for comparison
- Higher numbers = less sustainable

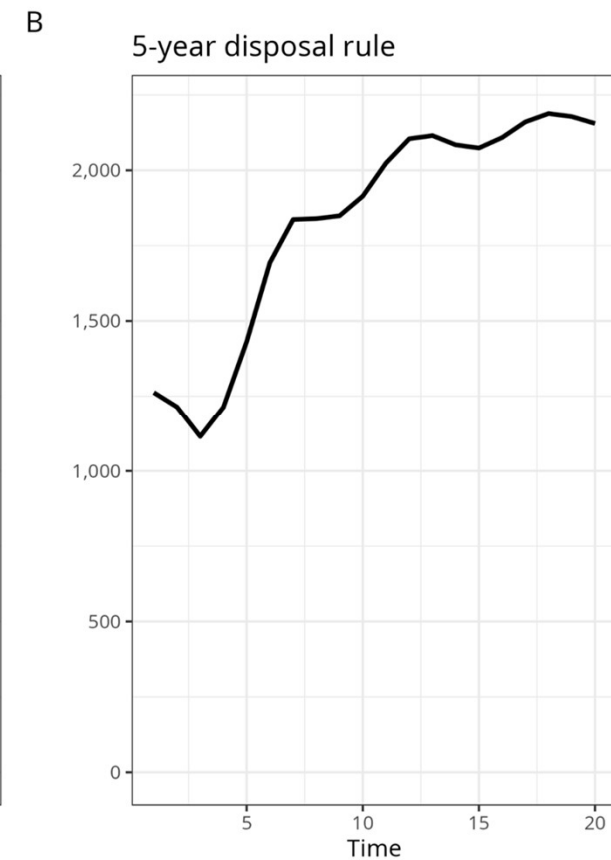
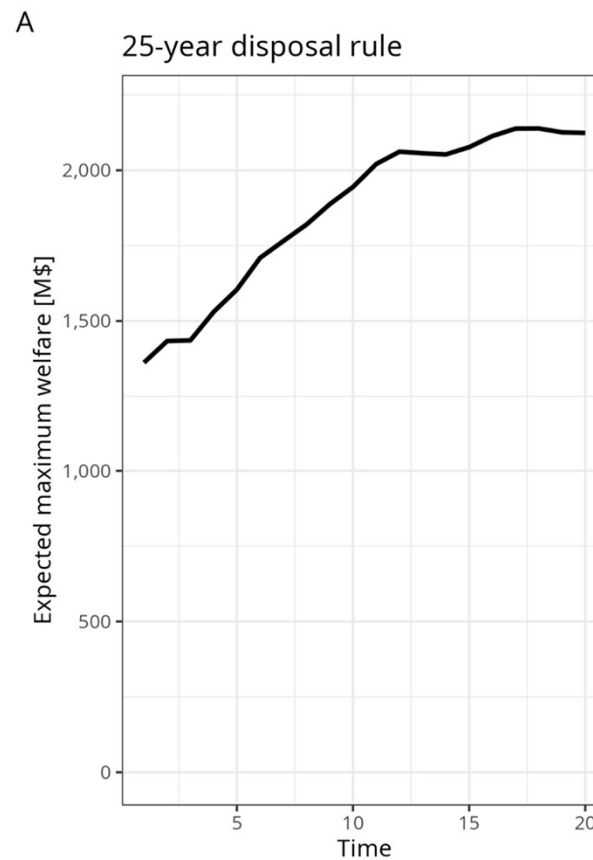


25- vs 5-year PMD (Economic welfare)

Model is uncalibrated – illustrative only

Expected maximum (upper bound) economic welfare to measure economic value to society from orbit use under policy

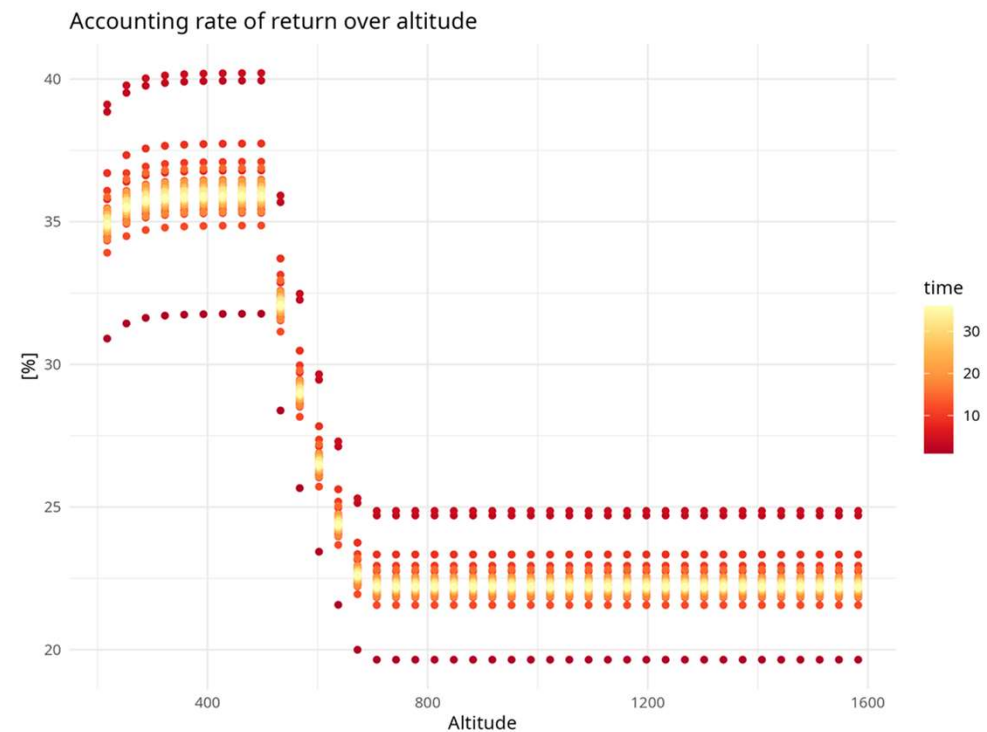
- Compute upper bound on expected NPV from fringe satellites
- Higher numbers = greater social value



Policy exercise: 5-year PMD vs 25-year w/ OUF

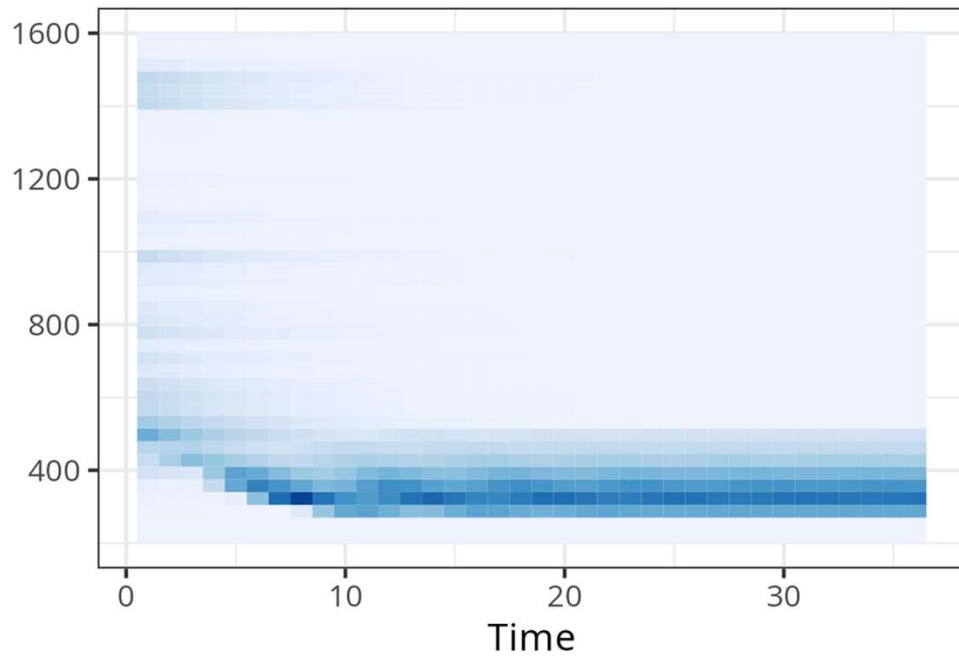
Are stricter disposal rules equivalent to taxes? Can assess in OPUS

- Two separate scenarios
- Start from July 2022 initial population
- Assume full compliance

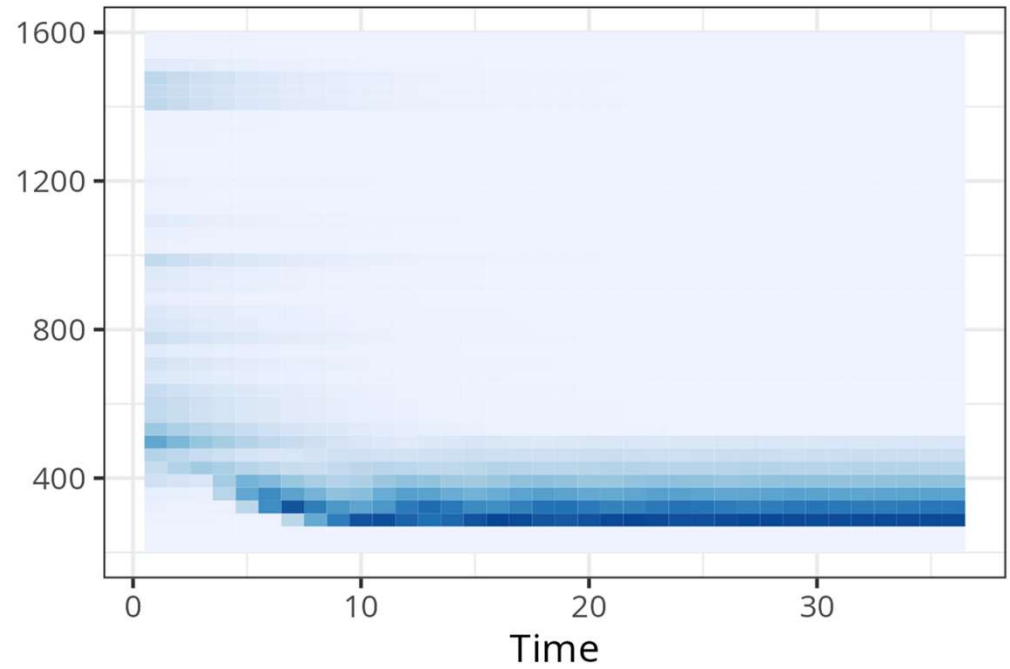


5-year PMD vs 25-year w/ OUF

5 year PMD

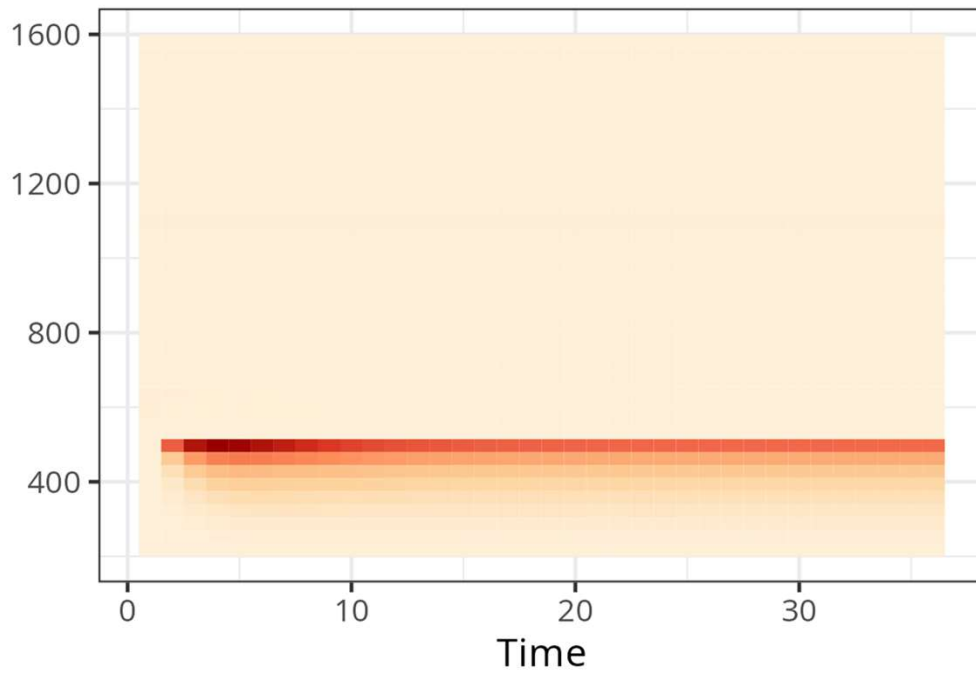


25 year PMD w/ OUF

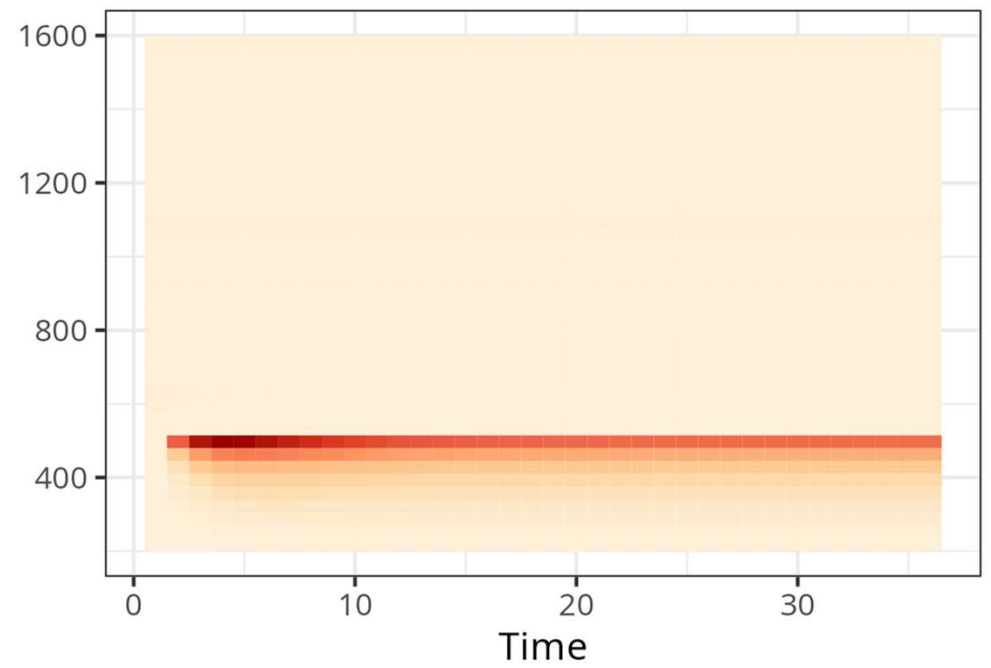


5-year PMD vs 25-year w/ OUF

5 year PMD



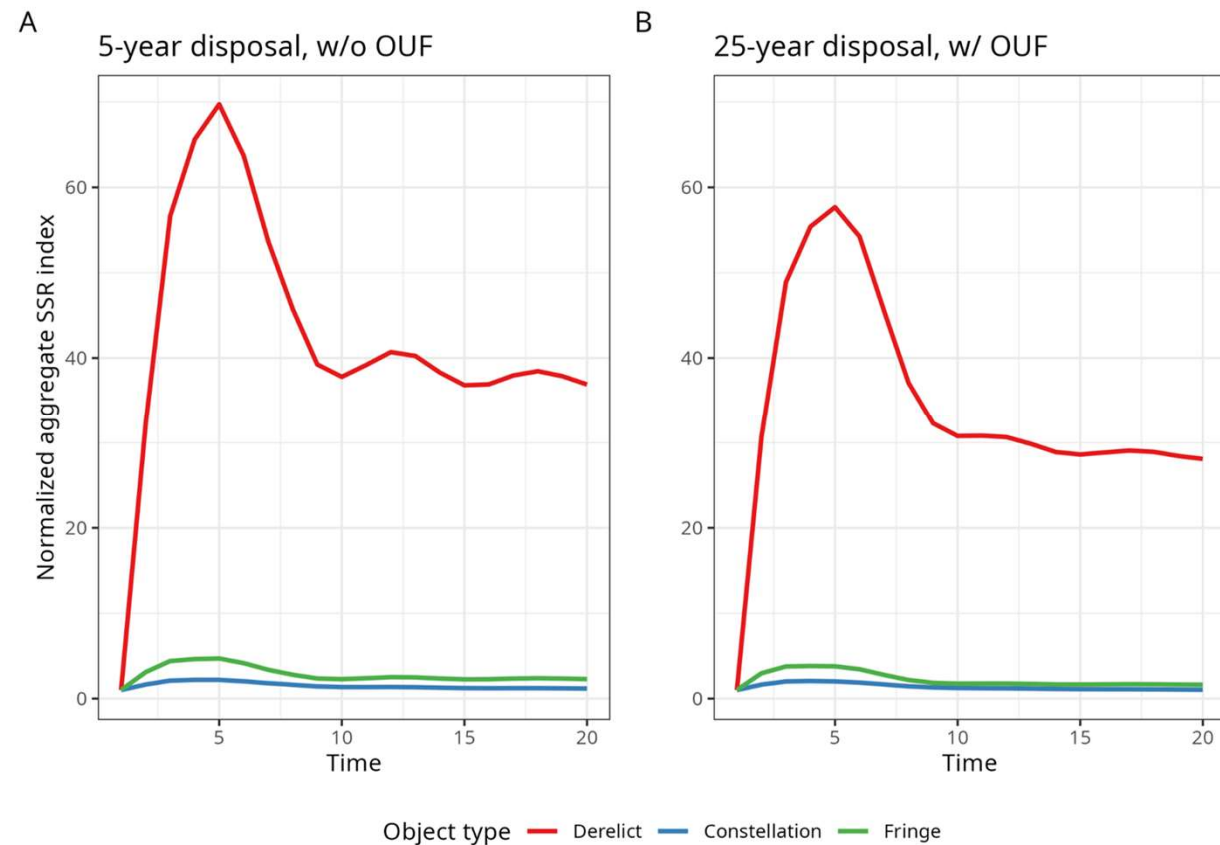
25 year PMD w/ OUF



5-year PMD vs 25-year w/ OUF (SSR)

Model is uncalibrated – illustrative only

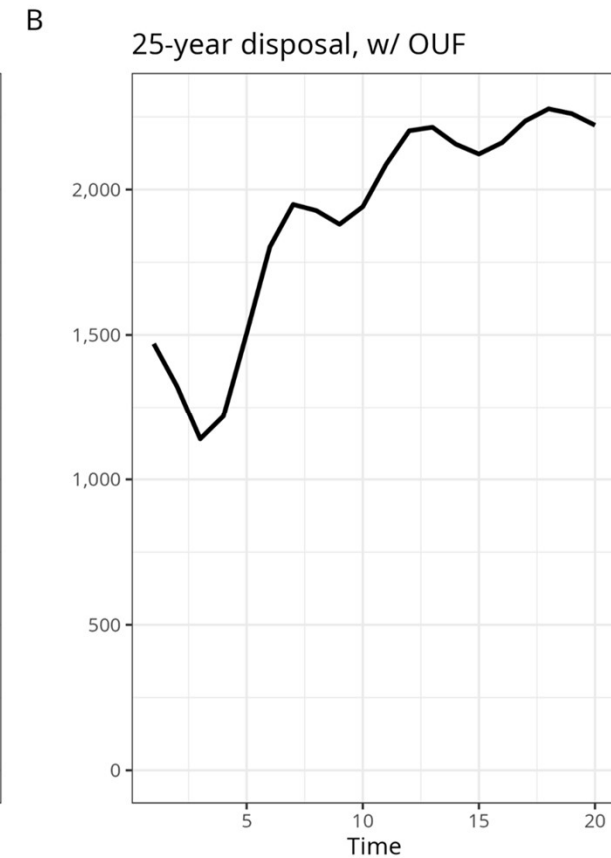
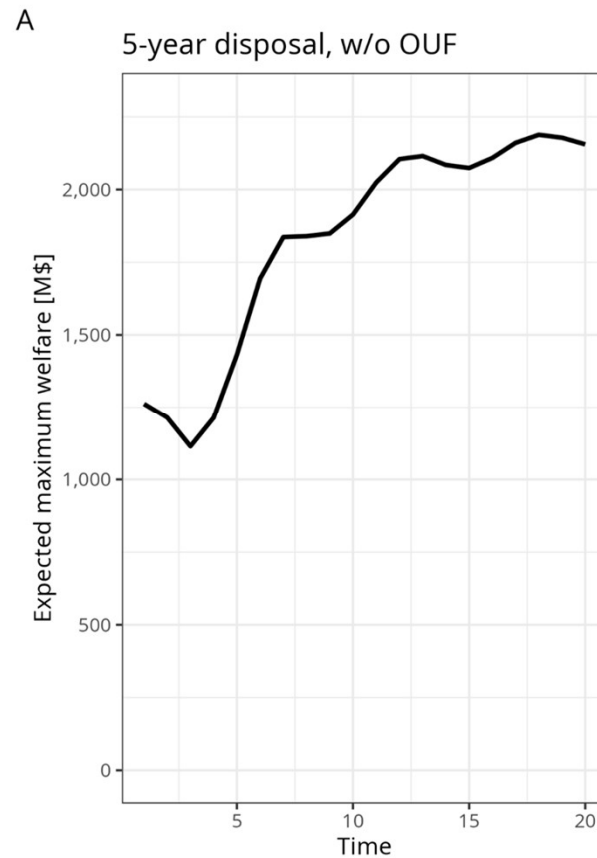
- Take with salt
- *In principle*, OUF can replicate stricter disposal timeline



5-year PMD vs 25-year w/ OUF (Economic welfare)

Model is uncalibrated – illustrative only

- Take with salt
- *In principle*, OUF can replicate stricter disposal timeline



Future R&D directions

Some directions for improvements:

- Improved economic parameter calibration
- Models of launch capacity and prices
- More detailed models of operator behavior
- Models of constellation behavior
- Models of non-commercial demand for satellite services

